

Implementation of ARINC 429 Controller using VIVADO

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Abstract: ARINC 429 is a widely used data communication standard in avionics systems, providing a robust and reliable means of exchanging information between aircraft subsystems. The signals obtained from the other electronic equipment in the avionics system should be converted to the ARINC 429 protocol standard. To convert the signals, an controller is designed to handle both transmission and reception of data over ARINC 429 buses, offering capabilities such as data encoding/decoding, error checking, and scheduling of messages. The proposed ARINC 429 controller is implemented using Verilog language and the functionality is verified using tool Vivado 2023.1v. This controller provides a versatile solution for avionics applications, enhancing the interoperability of avionics equipment and contributing to the overall safety and reliability of aircraft systems.

Keywords: ARINC 429, Vivado, VLSI.

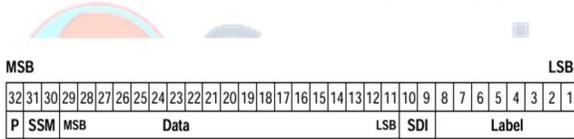
I. INTRODUCTION

ARINC 429 is a communication standard commonly employed in avionics to facilitate data exchange between different electronic systems within an aircraft[1]. It was developed by Aeronautical Radio, Inc. (ARINC) to provide a reliable, straightforward, and standardized method of transferring information within avionics systems. ARINC 429 uses a one-way communication protocol. Data is sent in a unidirectional manner from a single transmitter to multiple receivers. ARINC 429 employs twisted-pair cables and differential

signaling for noise immunity. It uses a bipolar Return-to-Zero (RZ) modulation. Two data rates are supported : Low-speed: 12.5 kbps, High-speed: 100 kbps. A parity bit is utilized to detect errors.. ARINC 429 supports continuous or periodic data transmission, with a self-clocking scheme. Signals are transmitted with a voltage differential of $\pm 10V$, ensuring robust communication even in noisy environments.

II DATA FORMAT

ARINC 429 is a standard data communication protocol used in avionics systems to ensure reliable data exchange between aircraft equipment[2][3]. The protocol uses a 32-bit data word format for communication. The first bit is reserved for parity, implementing odd parity for error detection. Bits 2 to 10 represent the Source/Destination Identifier, which identifies the transmitting or intended receiving system. The core data is contained in bits 11 to 29, providing 18 bits of actual information. Bits 30 and 31 form the Sign/Status Matrix (SSM), which offers supplementary status information about the data's validity or condition. Finally, the Label field, encoded in bits 32, is a unique identifier in octal format, categorizing the type of data being transmitted. This structured format ensures clarity, reliability, and efficient communication in avionics systems[4].



ARINC 429 32-bit Word Format
Fig 1 : ARINC 429 32 – bit Data Format

III METHODOLOGY

The above Fig 2 represents the block diagram of ARINC 429 controller where the inputs are of 32 – bits, the controller block consists of

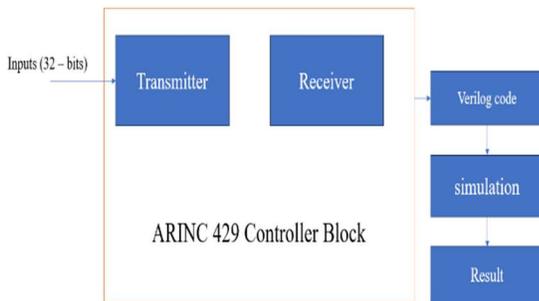


Fig 2 : General Block diagram of Implementation of ARINC 429 Controller

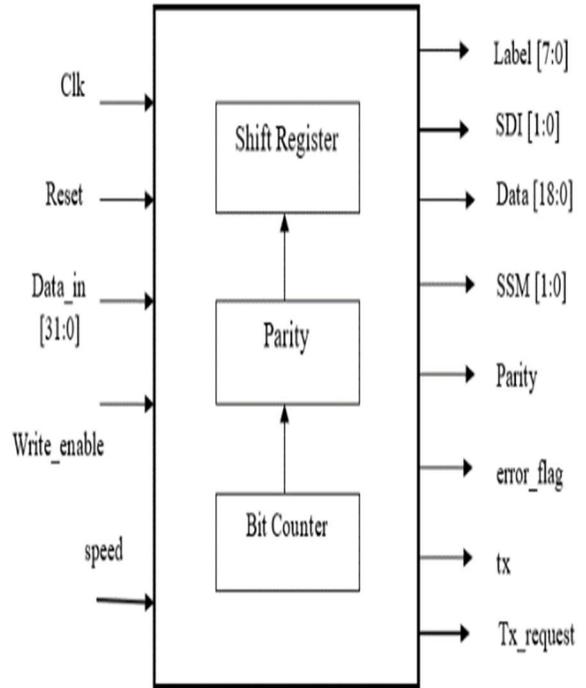


Fig 3 : Transmitter Block of ARINC 429 Controller

Transmitter and Receiver block. The functionality is coded using Verilog language and simulation is verified and results are obtained[5].

The transmitter is initialized using reset input. The transmitter module's shift register transforms the test bench's 32-bit parallel data into serial data. When parity is enabled, the parity bit is also included to the ARINC data. In order to ensure proper data receipt, ARINC uses odd parity as an error check. The Label is transmitted first, that is MSB , followed by the remainder of the bit field, that is LSB, when transmitting data words on the ARINC bus. The Label is always communicated first, followed by the remainder of the ARINC word, in reverse order. This ensures compatibility with legacy systems. The receiving unit is in charge of translating data and rearranging bits into the correct sequence. This is illustrated in Fig 3.

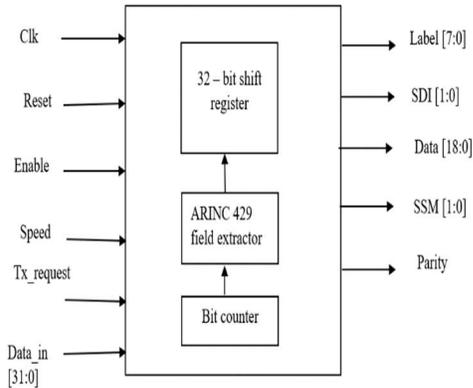


Fig 4 : Receiver Block of ARINC 429 Controller

The receiver listens for messages transmitted over the bus but does not send any information back. This simplifies the communication process and reduces system complexity. The primary function of the ARINC 429 receiver is to receive, decode, and process the 32-bit data words transmitted over the bus. The receiver constantly listens to the differential signals transmitted on the ARINC 429 bus. It is designed to handle signal reception at 100 Hz or 400 Hz baud rates (standard transmission speeds). The receiver processes the 32-bit word by extracting its individual components, with the Label indicating the type of data (e.g., altitude, airspeed, etc.).

The Data part contains the actual value being transmitted (e.g., numerical data like the airspeed). The Parity bit is checked to ensure that no errors have occurred during transmission. The ARINC 429 receiver includes error-detection capabilities to ensure data integrity. The parity bit is used to check whether an error has occurred during transmission. If a received word does not match the expected label or format, the receiver can discard or flag the word as invalid. Errors such as bit errors, framing errors, and sync errors can be detected by the receiver. If an error is detected, it may flag the

data or request a retransmission. The receiver typically only processes data that matches its expected label. Each system on the ARINC 429 bus can have a different label associated with the data it sends. For instance, one receiver might only accept data with a label of "22" (airspeed), while another might accept a label of "19" (altitude). As shown in Fig 4.

IV IMPLEMENTATION

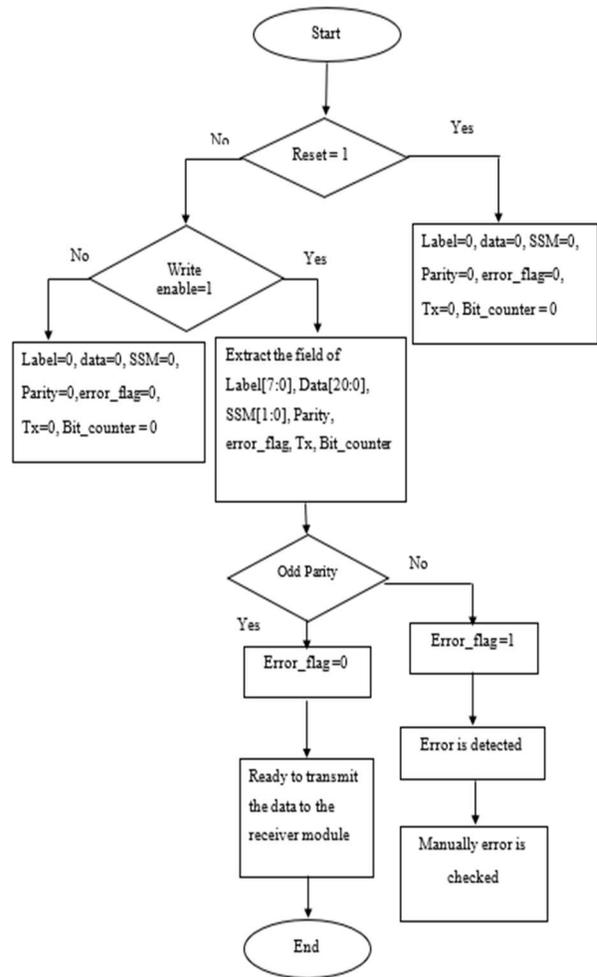


Fig 5 : Flow chart of ARINC 429 Transmitter

The provided flow chart as shown in Fig 5 outlines the operational sequence of a transmitter module. It commences by ascertaining whether a reset condition exists, and if so, it sets the reset flag to 1. Subsequently, it initializes several variables:



Fig 9 : Simulation of ARINC 429 Receiver with Parity error



Fig 10 : Simulation of ARINC 429 Receiver without parity error

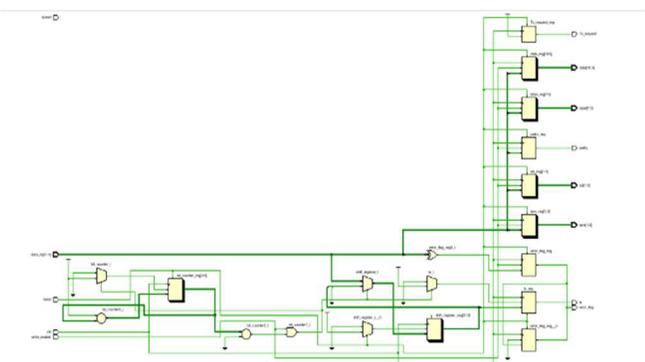


Fig 11 : RTL Schematic of ARINC 429 Transmitter

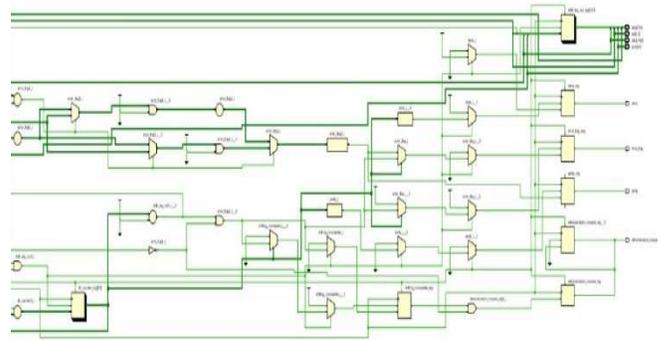


Fig 12 : RTL Schematic of ARINC 429 Receiver

VI. CONCLUSION

The ARINC 429 receiver is a vital part of the avionics systems used in modern aircraft, ensuring that critical flight data is received, decoded, and processed accurately. It plays a key role in maintaining the integrity and performance of aviation systems by ensuring reliable data transmission, error detection, and synchronization. Through its robust error-checking and data validation capabilities, it supports safe and efficient aircraft operations in the highly demanding aerospace environment.

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